



Interval Methods for Robust Variable-Structure Control with One- and Two-Sided State Constraints

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A. Rauh et al.: Interval Methods for Robust Variable-Structure Control with One- and Two-Sided State Constraints

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Contents

- First- and second-order sliding mode control for systems in nonlinear controller canonical form (restricted to the single-input single-output case)
- Extensions to dynamic systems with interval uncertainty
- Consideration of one-sided state constraints (upper bounds on selected state/ output variables)
- Two-sided state constraints
- Simulation results
 - Control of an accelerated mass with non-negligible actuator dynamics
 - Control of the non-stationary thermal behavior of Solid Oxide Fuel Cell modules (with further input range and input rate constraints)
 - Control of an inverted pendulum
- Conclusions and outlook on future work

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First-Order vs. Second-Order Sliding Mode Control (1) System in nonlinear controller canonical form

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} \dot{x}_1(t) & \dots & \dot{x}_{n-1}(t) & \dot{x}_n(t) \end{bmatrix}^T$$
$$= \begin{bmatrix} x_2(t) & \dots & x_n(t) & a\left(\mathbf{x}(t), \mathbf{p}\right) + b\left(\mathbf{x}(t), \mathbf{p}\right) \cdot v(t) \end{bmatrix}^T$$

with the state vector $\mathbf{x}(t) \in \mathbb{R}^n$

Requirement for controllability

 $b\left(\mathbf{x}(t),\mathbf{p}
ight)
eq 0$ for any possible operating point and system parameter

Feedback linearizing control law for the output $y(t) = x_1(t)$

$$v(t) = \frac{-a\left(\mathbf{x}(t), \mathbf{p}\right) + u(t)}{b\left(\mathbf{x}(t), \mathbf{p}\right)} \in \mathbb{R}$$

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First-Order vs. Second-Order Sliding Mode Control (1)

System in nonlinear controller canonical form

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} \dot{x}_1(t) & \dots & \dot{x}_{n-1}(t) & \dot{x}_n(t) \end{bmatrix}^T$$
$$= \begin{bmatrix} x_2(t) & \dots & x_n(t) & a\left(\mathbf{x}(t), \mathbf{p}\right) + b\left(\mathbf{x}(t), \mathbf{p}\right) \cdot v(t) \end{bmatrix}^T$$

with the state vector $\mathbf{x}(t) \in \mathbb{R}^n$

Feedback linearizing control law for the output $y(t) = x_1(t)$

$$v(t) = \frac{-a\left(\mathbf{x}(t), \mathbf{p}\right) + u(t)}{b\left(\mathbf{x}(t), \mathbf{p}\right)} \in \mathbb{R}$$

System becomes a pure integrator chain of length n for perfect system and state information (trivially differentially flat system)

Sliding Mode Control First-Order vs. Second-Order Sliding Mode Control (2) *n*-th order integrator chain model with the output $y(t) = x_1(t)$ $\begin{vmatrix} \dot{x}_1(t) \\ \vdots \\ \dot{x}_{n-1}(t) \\ \dot{x}_n(t) \end{vmatrix} = \begin{vmatrix} x_2(t) \\ \vdots \\ x_n(t) \\ u(t) \end{vmatrix}$

Definition of the tracking error and its r-th time derivative

$$ilde{\xi}_1^{(r)}(t) = x_1^{(r)}(t) - x_{1,\mathrm{d}}^{(r)}(t) \qquad ext{with} \qquad r \in \{0,1,\dots,n\}$$

First-order sliding mode (Hurwitz polynomial of order n-1)

$$s := s(t) = \sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t) , \quad \alpha_{n-1} = 1 \qquad \Longrightarrow \qquad s \to 0$$

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Sliding Mode Control First-Order vs. Second-Order Sliding Mode Control (2) *n*-th order integrator chain model with the output $y(t) = x_1(t)$ $\begin{vmatrix} \dot{x}_1(t) \\ \vdots \\ \dot{x}_{n-1}(t) \\ \dot{x}_n(t) \end{vmatrix} = \begin{vmatrix} x_2(t) \\ \vdots \\ x_n(t) \\ u(t) \end{vmatrix}$

Definition of the tracking error and its r-th time derivative

$$ilde{\xi}_1^{(r)}(t) = x_1^{(r)}(t) - x_{1,\mathrm{d}}^{(r)}(t) \qquad ext{with} \qquad r \in \{0,1,\dots,n\}$$

Second-order sliding mode (integral component for $\alpha_{-1} \neq 0$)

$$\gamma_1 \dot{s} + \gamma_0 s = \alpha_{-1} \int_0^t \tilde{\xi}_1(\tau) d\tau + \sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t) , \quad \begin{array}{c} \gamma_0 > 0\\ \gamma_1 > 0 \end{array} \implies \begin{array}{c} s \to 0\\ \dot{s} \to 0 \end{array}$$

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Derivation of the Control Law (1)

Lyapunov function candidate (first-order sliding mode)

$$V^{\langle \mathrm{I}\rangle} = \frac{1}{2}s^2 > 0 \quad \text{for} \quad s \neq 0$$

Lyapunov function candidate (second-order sliding mode)

$$V^{\langle {
m II}
angle} = rac{1}{2} \cdot \left(s^2 + \lambda \dot{s}^2
ight) > 0 \quad {
m with \ the \ scaling \ factor} \quad \lambda > 0$$

$$\dot{V}^{\langle \mathbf{I} \rangle} = s \cdot \dot{s} = \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t)\right) \cdot \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r+1)}(t)\right) < 0 \quad \text{for} \quad s \neq 0$$

Stability requirement (second-order sliding mode), $\lambda=\gamma_1>0$

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Derivation of the Control Law (2)

Stability requirement (first-order sliding mode)

$$\dot{V}^{\langle \mathbf{I} \rangle} = s \cdot \dot{s} = \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t)\right) \cdot \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r+1)}(t)\right) < -\eta \cdot |s|$$

Stability requirement (second-order sliding mode), $\lambda=\gamma_1>0$

$$\dot{V}^{\langle \text{II} \rangle} = s \cdot \dot{s} + \dot{s} \cdot \left(-\gamma_0 \dot{s} + \sum_{r=0}^{n-1} \alpha_{r-1} \tilde{\xi}_1^{(r)}(t) + \alpha_{n-1} \cdot \left(u(t) - x_{1,d}^{(n)}(t) \right) \right) < 0$$

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Derivation of the Control Law (2)

Stability requirement (first-order sliding mode)

$$\left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t)\right) \cdot \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r+1)}(t)\right) < -\eta \cdot \left(\sum_{r=0}^{n-1} \alpha_r \tilde{\xi}_1^{(r)}(t)\right) \cdot \operatorname{sign}\left(s\right)$$

Stability requirement (second-order sliding mode), $\lambda = \gamma_1 > 0$

$$\dot{V}^{\langle \mathrm{II} \rangle} < -\eta_1 \cdot |\dot{s}| - \eta_2 \cdot |s| \cdot |\dot{s}| = -\dot{s} \cdot \mathrm{sign}\left(\dot{s}\right) \cdot \left(\eta_1 + \eta_2 \cdot |s|\right)$$

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Derivation of the Control Law (3)

Control law (first-order sliding mode)

$$u(t) = u^{\langle \mathbf{I} \rangle}(t) = x_{1,d}^{(n)}(t) - \sum_{r=0}^{n-2} \alpha_r \tilde{\xi}_1^{(r+1)}(t) - \tilde{\eta} \cdot \operatorname{sign}(s)$$

Questions

- What are necessary extensions for the interval case?
- What are the implementation requirements for an *interval-valued control signal*?
- Why/ How to generalize *preferably* the first-order case?

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Derivation of the Control Law (3)

Control law (second-order sliding mode)

$$u(t) = u^{\langle II \rangle}(t) = x_{1,d}^{(n)}(t) + \frac{1}{\alpha_{n-1}} \cdot \left(\gamma_0 \dot{s} - s - \sum_{r=0}^{n-1} \alpha_{r-1} \tilde{\xi}_1^{(r)}(t) - \operatorname{sign}(\dot{s}) \cdot (\tilde{\eta}_1 + \tilde{\eta}_2 \cdot |s|) \right)$$

Questions

- What are necessary extensions for the interval case?
- What are the implementation requirements for an *interval-valued control signal*?
- Why/ How to generalize *preferably* the first-order case?

Interval-Based Sliding Mode Control (1)

Definition of tracking error signals and sliding surface

- $\bullet\,$ Specification of a sufficiently smooth desired output trajectory $y_{\rm d}=x_{1,{\rm d}}$
- Interval definition of the tracking error and its derivatives

$$\tilde{\xi}_1^{(r)} \in \left[\tilde{\xi}_1^{(r)}\right] = \left[x_1^{(r)}\right] - x_{1,d}^{(r)}, \ r \in \{0, 1, \dots, n\}$$

• As before: Desired operating points are located on the *sliding surface* $s:=\tilde{\xi}_1^{(n-1)}(t)+\sum_{r=0}^{n-2}\alpha_r\cdot\tilde{\xi}_1^{(r)}(t)=0$

• $\alpha_0, \ldots, \alpha_{n-2}$ are coefficients of a Hurwitz polynomial of order n-1

Interval-Based Sliding Mode Control (1)

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• As before: Desired operating points are located on the sliding surface $s:=\tilde{\xi}_1^{(n-1)}(t)+\sum_{r=0}^{n-2}\alpha_r\cdot\tilde{\xi}_1^{(r)}(t)=0$

Guaranteed stabilizing control: Lyapunov function candidate $V=\frac{1}{2}s^2>0 \quad \text{with} \quad \dot{V}=s\cdot\dot{s}<0 \quad \text{for} \quad s\neq 0$

Interval-Based Sliding Mode Control (2)

Guaranteed stabilization despite uncertainty: Interval formulation of a variable-structure control law

$$\left[v^{\langle \mathbf{I} \rangle}\right] := \frac{-a\left([\mathbf{x}], [\mathbf{p}]\right) + x_{1, \mathbf{d}}^{(n)} - \sum_{r=0}^{n-2} \alpha_r \cdot \left[\tilde{\xi}_1^{(r+1)}\right] - \tilde{\eta} \cdot \operatorname{sign}\left([s]\right)}{b\left([\mathbf{x}], [\mathbf{p}]\right)}$$

with a suitably chosen parameter $\tilde{\eta} > 0$ and $0 \notin b\left([\mathbf{x}], [\mathbf{p}] \right)$

Guaranteed stabilizing control: Extraction of suitable point values

$$\mathcal{V} := \{ \underline{v} - \epsilon, \underline{v} + \epsilon, \overline{v} - \epsilon, \overline{v} + \epsilon \}$$

with $\underline{v} := \inf\{[v]\}$, $\overline{v} := \sup\{[v]\}$ and some small $\epsilon > 0 \Longrightarrow \dot{V} < 0$ needs to be satisfied with certainty

Interval-Based Sliding Mode Control (2)

Guaranteed stabilization despite uncertainty: Interval formulation of a variable-structure control law

$$\left[v^{\langle \mathbf{I}\rangle}\right] := \frac{-a\left([\mathbf{x}], [\mathbf{p}]\right) + x_{1, \mathbf{d}}^{(n)} - \sum_{r=0}^{n-2} \alpha_r \cdot \left[\tilde{\xi}_1^{(r+1)}\right] - \tilde{\eta} \cdot \operatorname{sign}\left([s]\right)}{b\left([\mathbf{x}], [\mathbf{p}]\right)}$$

with a suitably chosen parameter $\tilde{\eta}>0$ and $0\not\in b\left([\mathbf{x}],[\mathbf{p}]\right)$

Guaranteed stabilizing control: Extraction of suitable point values

- Guaranteed stabilization of system dynamics
- Extension: Guaranteed state constraints in terms of strict one- and two-sided barrier functions
- Inclusion of bounds on input variation rates (reduction of the effect of chattering)

Sliding Mode Control with One-Sided State Constraints (1)

Specification of an upper state constraint

$$x_1 < \bar{x}_{1,\max} := x_{1,d} + \Delta x_{1,\max}$$
 with $\Delta x_{1,\max} > 0$

Extension of the Lyapunov function candidate by a one-sided barrier function (repelling potential)

Extended ansatz for a Lyapunov function candidate

$$V^{\langle j, \mathbf{A} \rangle} = V^{\langle j \rangle} + V^{\langle \mathbf{A} \rangle} > 0$$
 for $s \neq 0$ with

$$V^{\langle A \rangle} = \rho_{V} \cdot \ln \left(\frac{\sigma_{V} \cdot \bar{x}_{1,\max}}{\bar{x}_{1,\max} - x_{1}} \right) \text{ and } x_{1} < \bar{x}_{1,\max} \text{ for both } j \in \{I, II\}$$

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Sliding Mode Control with One-Sided State Constraints (1)

Specification of an upper state constraint

$$x_1 < ar{x}_{1,\max} := x_{1,\mathrm{d}} + \Delta x_{1,\max}$$
 with $\Delta x_{1,\max} > 0$

Computation of the corresponding time derivative Extended ansatz for a Lyapunov function candidate

$$\begin{split} \dot{V}^{\langle j, \mathbf{A} \rangle} &= \dot{V}^{\langle j \rangle} + \dot{V}^{\langle \mathbf{A} \rangle} < 0 \quad \text{with} \\ \dot{V}^{\langle \mathbf{A} \rangle} &= \frac{\rho_{\mathbf{V}}}{\bar{x}_{1, \max}} \cdot \left(\frac{-x_1 \cdot \dot{\bar{x}}_{1, \max} + \dot{x}_1 \cdot \bar{x}_{1, \max}}{\bar{x}_{1, \max} - x_1} \right) \ , \quad \rho_{\mathbf{V}} > 0 \ , \quad \sigma_{\mathbf{V}} > 0 \end{split}$$

Note

Dominating influence of $\dot{V}^{\langle j\rangle}$ in the neighborhood of s=0 must be given



Sliding Mode Control with One-Sided State Constraints (1)

Specification of an upper state constraint

 $x_1 < \bar{x}_{1,\max} := x_{1,\mathrm{d}} + \Delta x_{1,\max}$ with $\Delta x_{1,\max} > 0$

Modified stability requirement (first-order case)

$$s \cdot \underbrace{\left(\sum_{r=0}^{n-2} \alpha_r \tilde{\xi}_1^{(r+1)} + u - x_{1,\mathrm{d}}^{(n)} + \eta \cdot \operatorname{sign}\left(s\right) + \frac{1}{s} \cdot \dot{V}^{\langle \mathrm{A} \rangle}\right)}_{-\beta \cdot \operatorname{sign}(s)} < 0$$

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Sliding Mode Control with One-Sided State Constraints (2)

New control signal for the first-order case

$$u = u^{\langle \mathbf{I}, \mathbf{A} \rangle} = u^{\langle \mathbf{I} \rangle} - \frac{s}{s^2 + \tilde{\epsilon}} \cdot \dot{V}^{\langle \mathbf{A} \rangle}$$

New control signal for the second-order case

$$u = u^{\langle \mathrm{II}, \mathrm{A} \rangle} = u^{\langle \mathrm{II} \rangle} - \frac{1}{\alpha_{n-1}} \cdot \frac{\dot{s}}{\dot{s}^2 + \tilde{\epsilon}} \cdot \dot{V}^{\langle \mathrm{A} \rangle}$$

Note

- The approximations $\frac{1}{s} \approx \frac{s}{s^2 + \tilde{\epsilon}}$ and $\frac{1}{\dot{s}} \approx \frac{\dot{s}}{\dot{s}^2 + \tilde{\epsilon}}$ are only necessary for $|s| \gg 0$ and $|\dot{s}| \gg 0$.
- The variable-structure part is deactivated in a close vicinity of |s| = 0 and |s| = 0 in the following interval case. For 0 ∈ [s], 0 ∈ [s], the sign of s and s cannot be determined unambiguously.

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Sliding Mode Control with Two-Sided State Constraints (1)

Specification of worst-case state deviations

$$|x_1 - x_{1,d}| \ge \bar{\chi} > 0$$
, $\bar{\chi} = \text{const}$

Extension of the Lyapunov function candidate by a two-sided barrier function

Extended ansatz for a Lyapunov function candidate

$$V^{\langle j,{
m B}
angle}=V^{\langle j
angle}+V^{\langle {
m B}
angle}>0 \ \ \ {
m for} \ \ s
eq 0 \ \ \ {
m with}$$

$$V^{\langle \mathrm{B} \rangle} = \rho_{\mathrm{V}} \cdot \ln \left(\frac{\bar{\chi}^{2l}}{\bar{\chi}^{2l} - (x_1 - x_{1,\mathrm{d}})^{2l}} \right) \text{ and } l \in \mathbb{N} \text{ for both } j \in \{\mathrm{I},\mathrm{II}\}$$

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Sliding Mode Control with Two-Sided State Constraints (1)

Specification of worst-case state deviations

$$|x_1 - x_{1,d}| \ge \bar{\chi} > 0$$
, $\bar{\chi} = \text{const}$

Computation of the corresponding time derivative Extended ansatz for a Lyapunov function candidate

$$\begin{split} \dot{V}^{\langle j,\mathrm{B}\rangle} &= \dot{V}^{\langle j\rangle} + \dot{V}^{\langle \mathrm{B}\rangle} < 0 \quad \text{with} \\ \dot{V}^{\langle \mathrm{B}\rangle} &= \rho_{\mathrm{V}} \cdot \frac{2l \cdot \left(x_1 - x_{1,\mathrm{d}}\right)^{2l-1} \cdot \left(\dot{x}_1 - \dot{x}_{1,\mathrm{d}}\right)}{\bar{\chi}^{2l} - \left(x_1 - x_{1,\mathrm{d}}\right)^{2l}} \ , \quad \rho_{\mathrm{V}} > 0 \end{split}$$

Note

Dominating influence of $\dot{V}^{\langle j \rangle}$ in the neighborhood of s = 0 must be given

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Sliding Mode Control with Two-Sided State Constraints (2)

New control signal for the first-order case

$$u^{\langle \mathbf{I},\mathbf{B}\rangle} = u^{\langle \mathbf{I}\rangle} - s^{-1} \cdot \dot{V}^{\langle \mathbf{B}\rangle}$$

New control signal for the second-order case

$$u^{\langle \mathrm{II},\mathrm{B}\rangle} = u^{\langle \mathrm{II}\rangle} - (\alpha_{n-1} \cdot \dot{s})^{-1} \cdot \dot{V}^{\langle \mathrm{B}\rangle}$$

Note

The same approximations $\frac{1}{s} \approx \frac{s}{s^2 + \tilde{\epsilon}}$ and $\frac{1}{\dot{s}} \approx \frac{\dot{s}}{\dot{s}^2 + \tilde{\epsilon}}$ are necessary as before for $|s| \gg 0$ and $|\dot{s}| \gg 0$.

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Interval-Based Sliding Mode Control (continued)

Guaranteed stabilization despite uncertainty: Interval formulation of a variable-structure control law

$$\left[v^{\langle \mathbf{I} \rangle}\right] := \frac{-a\left([\mathbf{x}], [\mathbf{p}]\right) + x_{1, \mathbf{d}}^{(n)} - \sum_{r=0}^{n-2} \alpha_r \cdot \left[\tilde{\xi}_1^{(r+1)}\right] - \tilde{\eta} \cdot \operatorname{sign}\left([s]\right)}{b\left([\mathbf{x}], [\mathbf{p}]\right)}$$

Extension in the case of one-sided state constraints

$$\left[v^{\langle \mathrm{I},\mathrm{A}\rangle}\right] = \left[v^{\langle \mathrm{I}\rangle}\right] - \frac{1}{b\left([\mathbf{x}],[\mathbf{p}]\right)} \cdot \frac{[s]}{[s]^2 + \tilde{\epsilon}} \cdot \left[\dot{V}^{\langle \mathrm{A}\rangle}\right]$$

Interval-Based Sliding Mode Control (continued)

Guaranteed stabilization despite uncertainty: Interval formulation of a variable-structure control law

$$\left[v^{\langle \mathbf{I}\rangle}\right] := \frac{-a\left([\mathbf{x}], [\mathbf{p}]\right) + x_{1, \mathbf{d}}^{(n)} - \sum_{r=0}^{n-2} \alpha_r \cdot \left[\tilde{\xi}_1^{(r+1)}\right] - \tilde{\eta} \cdot \operatorname{sign}\left([s]\right)}{b\left([\mathbf{x}], [\mathbf{p}]\right)}$$

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Extension in the case of two-sided state constraints

$$\left[v^{\langle \mathbf{I},\mathbf{B}\rangle}\right] = \left[v^{\langle \mathbf{I}\rangle}\right] - \frac{1}{b\left([\mathbf{x}],[\mathbf{p}]\right)} \cdot \frac{[s]}{[s]^2 + \tilde{\epsilon}} \cdot \left[\dot{V}^{\langle \mathbf{B}\rangle}\right]$$

Example 1		Outlook	
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Velocity Control of a Point Mass

System model

- Position: x_1
- Velocity: x_2
- Input force: x_3 (mass normalized to 1)

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} x_2 \\ x_3 \\ p_1 x_1 + p_2 x_2 + p_3 x_3 + p_4 v \end{bmatrix}$$

with the uncertain parameters $p_i \in [-0.1; 0.1]$, $i \in \{1, 2, 3\}$, and $p_4 = 1$, containing both asymptotically stable and unstable realizations

Example 1 000000		Outlook 000	

Control Parameterizations

	Lyapunov function V	Barrier	System parameters	Measurement tolerance for x_1	Variable- structure gain
Case 1	$V^{\langle { m I} angle}$,	—	$p_i = 0,$	—	$\tilde{\eta} = 20$
	$\alpha_0 = 1$,		$i = \{1, 2, 3\},\$		
	$\alpha_1 = 0.9$		$p_4 = 1$		
Case 2	$V^{\langle \mathrm{I},\mathrm{A} angle}$,	$ \rho_{\rm V} = 0.5, $	$p_i = 0$,	—	$\tilde{\eta} = 20$
	$\alpha_0 = 1$,	$\sigma_{\mathrm{V}} = 1$,	$i = \{1, 2, 3\},\$		
	$\alpha_1 = 0.9$	$\Delta x_{1,\max} = 0.01$	$p_4 = 1$		
Case 3	$V^{\langle \mathrm{I},\mathrm{A} angle}$,	$ \rho_{\rm V} = 0.75, $	$p_i \in [-0.1 ; 0.1],$	$0.0025 \cdot [-1;1]$	$\tilde{\eta} = 20$
	$\alpha_0 = 15$,	$\sigma_{\mathrm{V}} = 1$,	$i = \{1, 2, 3\},\$		
	$\alpha_1 = 0.9$	$\Delta x_{1,\max} = 0.01$	$p_4 = 1$		
Case 4	$V^{\langle { m I} angle}$,	—	$p_i \in [-0.1; 0.1],$	$0.0025 \cdot [-1;1]$	$\tilde{\eta} = 400$
	$\alpha_0 = 15$,		$i = \{1, 2, 3\},\$		
	$\alpha_1 = 0.9$		$p_4 = 1$		
Case 5	$V^{\langle \mathrm{I},\mathrm{B} angle}$,	$\rho_{\rm V} = 5$,	$p_i \in [-0.1 ; 0.1],$	$0.0025 \cdot [-1;1]$	$\tilde{\eta} = 400$
	$\alpha_0 = 15$,	l = 1,	$i = \{1, 2, 3\},\$		
	$\alpha_1 = 0.9$	$\bar{\chi} = 0.165$	$p_4 = 1$		





Violation of one-sided state constraint \implies Barrier function is **deactivated**, parameters are assumed to be **exactly known**





No violation of one-sided state constraint \implies Barrier function is activated, parameters are assumed to be exactly known





No violation of one-sided state constraint \implies Barrier function is activated, parameters and measured states are uncertain





Violation of two-sided state constraint \implies Barrier function is **deactivated**, parameters and measured states are **uncertain**





No violation of two-sided state constraint \implies Barrier function is activated, parameters and measured states are uncertain

	Example 2	Outlook	
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Control-Oriented Modeling of SOFC Systems (1)

Configuration of the SOFC test rig at the Chair of Mechatronics

- Supply of fuel gas (hydrogen and/or mixture of methane, carbon monoxide, water vapor)
- Supply of air
- Independent preheaters for fuel gas and air
- Stack module containing fuel cells in electric series connection
- Electric load as disturbance



		Example 2		Outlook	
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Control-Oriented Modeling of SOFC Systems (2)

Spatial semi-discretization of the fuel cell stack module



Sliding Mode Control Example 1 Example 2 Example 3 Outlook Conclusions

Control-Oriented Modeling of SOFC Systems (3)

Mathematical representation of the piecewise homogeneous temperature distribution \implies spatial semi-discretization

$$\dot{\vartheta}_{\mathcal{I}}(t) = \frac{1}{c_{\mathcal{I}}m_{\mathcal{I}}} \left(\dot{Q}_{\mathrm{HT}}^{\mathcal{I}}(t) + \sum_{\mathrm{G} \in \{\mathrm{AG}, \mathrm{CG}\}} \dot{Q}_{\mathrm{G}, \mathcal{I}_{j}^{-}}^{\mathcal{I}}(t) + \dot{Q}_{\mathrm{R}}^{\mathcal{I}}(t) + \dot{Q}_{\mathrm{EL}}^{\mathcal{I}}(t) \right)$$

- HT: Heat transfer (heat conduction and convection)
- G: Enthalpy flows of supplied gases
- R: Exothermic reaction enthalpy
- EL: Ohmic losses



Control-Oriented Modeling of SOFC Systems (3)

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Control-Oriented Modeling of SOFC Systems (3)

Mathematical representation of the piecewise homogeneous temperature distribution \implies spatial semi-discretization

$$\dot{\vartheta}_{\mathcal{I}}(t) = \frac{1}{c_{\mathcal{I}}m_{\mathcal{I}}} \left(\dot{Q}_{\mathrm{HT}}^{\mathcal{I}}(t) + \sum_{\mathrm{G} \in \{\mathrm{AG}, \mathrm{CG}\}} \dot{Q}_{\mathrm{G}, \mathcal{I}_{j}^{-}}^{\mathcal{I}}(t) + \dot{Q}_{\mathrm{R}}^{\mathcal{I}}(t) + \dot{Q}_{\mathrm{EL}}^{\mathcal{I}}(t) \right)$$

- HT: Heat transfer (heat conduction and convection)
- G: Enthalpy flows of supplied gases
- R: Exothermic reaction enthalpy
- EL: Ohmic losses





Control-Oriented Modeling of SOFC Systems (4)

Local mass flow balances in the semi-discretized fuel cell stack module



Anode gas composition: $\dot{m}_{AG,in}(t) = \dot{m}_{H_2,in}(t) + \dot{m}_{N_2,in}(t) + \dot{m}_{H_2O,in}(t)$



Control-Oriented Modeling of SOFC Systems (4)

Local mass flow balances in the semi-discretized fuel cell stack module



Anode gas composition: Air $\dot{m}_{\rm CG,in}(t)$

	Example 2 000000000000000	Outlook 000	

Different Variants of the Finite Volume Model (1)



- Configuration (I): Typical for synthesizing a controller that is only applied during the system's heating phase
- Configuration (II): Simplest option for preventing local overtemperatures: Differentially flat or non-flat scenarios, depending on the choice of the system output $\vartheta_{\mathcal{I}^*}$
- Configuration (III): Generally non-flat configuration

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Different Variants of the Finite Volume Model (2)



System input: Cathode gas enthalpy flow (single-input single-output formulation) in configuration (II), preheater dynamics neglected

$$v_{\rm CG,in}(t) = \dot{m}_{\rm CG,in}(t) \cdot \vartheta_{\rm CG,in}(t)$$

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Different Variants of the Finite Volume Model (2)



System input: Cathode gas enthalpy flow (single-input single-output formulation) in configuration (II), preheater dynamics included

 $\boldsymbol{v_{\mathrm{CG,d}}(t)} = \dot{m}_{\mathrm{CG,d}}(t) \cdot \vartheta_{\mathrm{CG,d}}(t) \;, \; \dot{m}_{\mathrm{CG,d}}(t) \approx \dot{m}_{\mathrm{CG,in}}(t)$

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Different Variants of the Finite Volume Model (2)



Vector representation of the input (multi-input single-output formulation)

$$\mathbf{u}_{\mathrm{CG,d}}(t) = \begin{bmatrix} \dot{m}_{\mathrm{CG,d}}(t) \\ \vartheta_{\mathrm{CG,d}}(t) \end{bmatrix}$$

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Transformation into Nonlinear Controller Canonical Form (1)

Input-affine state-space representation

$$\dot{\mathbf{x}}(t) = \mathbf{f}\left(\mathbf{x}(t), \mathbf{p}, \frac{\mathbf{v}_{\text{CG,d}}(t)}{\mathbf{v}_{\text{AG,d}}(t)}\right)$$

Computation of Lie derivatives of the system output $y(t) = h(\mathbf{x}(t)) = \vartheta_{\mathcal{I}^*}, \ \mathbf{x}(t) \in \mathbb{R}^{\mathcal{N}}$

$$\frac{d^r y(t)}{dt^r} = y^{(r)}(t) = L_{\mathbf{f}}^r h(\mathbf{x}(t)) = L_{\mathbf{f}} \left(L_{\mathbf{f}}^{r-1} h(\mathbf{x}(t)) \right) , \quad r = 1, \dots, \delta - 1$$

with the relative degree $\boldsymbol{\delta}$ defined according to

$$\frac{\partial L_{\mathbf{f}}^{r}h(\mathbf{x}(t))}{\partial v_{\mathbf{CG},\mathbf{d}}} \equiv 0 \quad \text{for} \quad r = 0, \dots, \delta - 1 \quad \text{and} \quad \frac{\partial L_{\mathbf{f}}^{\delta}h(\mathbf{x}(t))}{\partial v_{\mathbf{CG},\mathbf{d}}} \neq 0$$

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Transformation into Nonlinear Controller Canonical Form (2)

Introduction of the new state vector

$$\boldsymbol{\xi} = \begin{bmatrix} h(\mathbf{x}), & L_{\mathbf{f}}h(\mathbf{x}), & \dots, & L_{\mathbf{f}}^{\delta-1}h(\mathbf{x}) \end{bmatrix}^T \in \mathbb{R}^{\delta} \text{ with } \xi_1 = y = h(\mathbf{x})$$

New set of state equations (Brunovsky canonical form)

$$\begin{bmatrix} \dot{\boldsymbol{\xi}}^T & \dot{\boldsymbol{\zeta}}^T \end{bmatrix}^T = \begin{bmatrix} L_{\mathbf{f}}h(\mathbf{x}), \dots, & L_{\mathbf{f}}^{\delta}h(\mathbf{x}) & L_{\mathbf{f}}^{\delta+1}h(\mathbf{x}), \dots, & L_{\mathbf{f}}^{\mathcal{N}}h(\mathbf{x}) \end{bmatrix}^T \\ = \begin{bmatrix} \xi_2, \dots, & \xi_{\delta}, & \tilde{a}(\mathbf{x}, \mathbf{p}, d) & \mathbf{a}^{\Diamond}(\mathbf{x}, \mathbf{p}, d)^T \end{bmatrix}^T \\ + \begin{bmatrix} 0, \dots, & \tilde{b}(\mathbf{x}, \mathbf{p}) \cdot \boldsymbol{v}_{\mathbf{CG}, \mathbf{d}} & \mathbf{b}^{\Diamond}(\mathbf{x}, \mathbf{p}, d, \boldsymbol{v}_{\mathbf{CG}, \mathbf{d}}, \dots)^T \end{bmatrix}^T \end{bmatrix}$$

with the additive bounded disturbance $d \in [d]$, $d \in \mathbb{R}$, and the interval parameters $\mathbf{p} \in [\mathbf{p}]$, $\mathbf{p} \in \mathbb{R}^{n_{\mathrm{p}}}$

Sliding Mode Control Example 1 Example 2 Example 3 Outlook Conclusions of Conclu

 $\begin{bmatrix} \dot{\boldsymbol{\xi}}^T & \dot{\boldsymbol{\zeta}}^T \end{bmatrix}^T = \begin{bmatrix} \xi_2, & \dots, & \xi_{\delta}, & \tilde{a}(\mathbf{x}, \mathbf{p}, d) & \mathbf{a}^{\Diamond}(\mathbf{x}, \mathbf{p}, d)^T \end{bmatrix}^T \\ + \begin{bmatrix} 0, & \dots, & \tilde{b}(\mathbf{x}, \mathbf{p}) \cdot \boldsymbol{v}_{\text{CG}, d} & \mathbf{b}^{\Diamond}(\mathbf{x}, \mathbf{p}, d, \boldsymbol{v}_{\text{CG}, d}, \dot{\boldsymbol{v}}_{\text{CG}, d}, \dots)^T \end{bmatrix}^T$

 $\bullet\,$ Use of the variable $v_{{\rm CG},{\rm d}}$ as the control input

• Derivation of an interval-based variable structure control law

Requirements

- Estimation of all state variables x, of the parameters p, the disturbance d, and their corresponding interval bounds in real time
- Note: If $\delta \equiv \mathcal{N}$, the output y coincides with the flat system output
- Otherwise: The bounded states ζ of the non-controllable internal dynamics act as disturbances onto the system model

- $\begin{bmatrix} \dot{\boldsymbol{\xi}}^T & \dot{\boldsymbol{\zeta}}^T \end{bmatrix}^T = \begin{bmatrix} \xi_2, & \dots, & \xi_\delta, & \tilde{a}(\mathbf{x}, \mathbf{p}, d) & \dot{\mathbf{a}}^{\Diamond}(\mathbf{x}, \mathbf{p}, d)^T \end{bmatrix}^T \\ + \begin{bmatrix} 0, & \dots, & \tilde{b}(\mathbf{x}, \mathbf{p}) \cdot \boldsymbol{v}_{\mathbf{CG}, \mathbf{d}} & \dot{\mathbf{b}}^{\Diamond}(\mathbf{x}, \mathbf{p}, d, \boldsymbol{v}_{\mathbf{CG}, \mathbf{d}}, \dot{\boldsymbol{v}}_{\mathbf{CG}, \mathbf{d}}, \dots)^T \end{bmatrix}^T$
- ullet Use of the variable $v_{{\rm CG},{\rm d}}$ as the control input
- Derivation of an interval-based variable structure control law

Possible estimation approaches

- Linear gain-scheduled state observer (Luenberger-like structure)
- Sensitivity-based estimation: Receding horizon approach (online minimization of a quadratic error measure)
- Observer in controller canonical form
- Robustification by linear matrix inequalities possible

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Handling of Input Rate Limitations

Extension of the system input by a further lag element

$$T_{\rm r} \cdot \dot{\tilde{v}}_{\rm CG,d} + \tilde{v}_{\rm CG,d} = \check{v}_{\rm CG,d}$$

with the new system input $\check{v}_{\mathrm{CG,d}}$ and the fixed time constant $T_\mathrm{r}>0$

Guaranteed compatibility of the actual system input with the rate constraints

$$\left|\dot{\tilde{v}}_{\mathrm{CG,d}}\right| \leq T_{\mathrm{r}}^{-1} \cdot \left(\sup\{[v_{\mathrm{CG,max}}]\} - \inf\{[v_{\mathrm{CG,max}}]\}\right)$$

under the prerequisite

$$\inf\{[v_{CG,max}]\} \equiv \inf\{[\tilde{v}_{CG,max}]\} \equiv \inf\{[\check{v}_{CG,max}]\},\\ \sup\{[v_{CG,max}]\} \equiv \sup\{[\tilde{v}_{CG,max}]\} \equiv \sup\{[\check{v}_{CG,max}]\}$$

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Control Parameterization: Basic Approach (Excerpt)

Yes Control signal fe	easible?	No	
Break , apply the control for the time step t_h , and proceed	Adaption of $\tilde{\eta}$ (Alternative: adapt the parameters α_r in definition of sliding surface)		
with the subsequent discretization step	Input saturation exceeded		
	a) $\tilde{v}_{CG,d}(t_k) < \inf\{\left[v_{CG,\max}\right]\}$	b) $\tilde{v}_{\mathrm{CG,d}}(t_k) > \sup\{\left[v_{\mathrm{CG,max}}\right]\}$	
	$\begin{split} & \frac{\mathbf{Increase} \ \tilde{\eta} \ \mathrm{if}}{\frac{\partial \tilde{v}_{\mathrm{CG,d}}}{\partial \tilde{\eta}}} > 0 \\ & \frac{\mathbf{Decrease} \ \tilde{\eta} \ \mathrm{if}}{\frac{\partial \tilde{v}_{\mathrm{CG,d}}}{\partial \tilde{\eta}}} < 0 \end{split}$	$\begin{aligned} & \frac{\mathbf{Increase } \tilde{\eta} \text{ if }}{\partial \tilde{v}_{\mathrm{CG,d}}} < 0 \\ & \frac{\partial \tilde{v}_{\mathrm{CG,d}}}{\partial \tilde{\eta}} < 0 \\ & \frac{\mathbf{Decrease } \tilde{\eta} \text{ if }}{\partial \tilde{v}_{\mathrm{CG,d}}} > 0 \end{aligned}$	

	Example 2	Outlook	
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Control Parameterization: Extension for Online Gain Scheduling

- Case 1: Offline parameterization with cutoff for control signal
- Case 2: Online parameterization
 - Define a desired eigenvalue λ_r of multiplicity $\delta 1$ on the sliding surface with corresponding parameters α_r
 - 2 Initialize $\tilde{\eta}$ with the desired value
 - ⁽³⁾ Adapt $\tilde{\eta}$ in a line-search approach (fixed number of $N_{\eta} = 5$ steps) to ensure compatibility of $\tilde{v}_{CG,d}$ with the control constraints
 - Stop, if admissible control is found
 - If no admissible control is found within N_{η} steps, adapt the eigenvalue λ_r and restart with Step (2); Break after at most $N_{\lambda} = 5$ repetitions
- Treatment of input rate constraints: Extension of the system input by a further lag element
- Simulation case study: L = N = 1, M = 3

	Example 2	Outlook	
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Simulation Results: Stack Temperatures



	Example 2	Outlook	
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Simulation Results: Tracking Error



	Example 2	Outlook	
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Simulation Results: CG Preheater Inputs





Control of an Inverted Pendulum

System model: Inverted pendulum on a carriage

• State vector
$$\mathbf{z} = egin{bmatrix} lpha & \dot{z} & \dot{lpha} \end{bmatrix}^T$$
; Control variable: Force $F_{ ext{C}}$

• Derivation of a set of first-order state equations with





Control of an Inverted Pendulum

System model: Inverted pendulum on a carriage

• State vector $\mathbf{z} = \begin{bmatrix} \alpha & z & \dot{\alpha} & \dot{z} \end{bmatrix}^T$; y: horizontal pendulum tip position

Derivation of a set of first-order state equations with



Sliding Mode Control Example 1 Example 2 Example 3 Outlook Conclusions

Approximate Transformation into Nonlinear Controller Canonical Form (1)

Successive computation of the output's Lie derivatives

$$\begin{split} y &= -a\sin(\alpha) + z , \quad \dot{y} = -a\cos(\alpha)\dot{\alpha} + \dot{z} \\ \ddot{y} &= a\sin(\alpha)\dot{\alpha}^2 + \frac{mg\sin(2\alpha) - 2ma\sin(\alpha)\dot{\alpha}^2 + 2F_{\rm C}}{2M + m\left(1 - \cos(2\alpha)\right)} \\ &- \cos\alpha\frac{2g\sin(\alpha)\left(M + m\right) - ma\dot{\alpha}^2\sin(2\alpha) + 2\cos(\alpha)F_{\rm C}}{2M + m\left(1 - \cos(2\alpha)\right)} \\ &\approx -p_1g\alpha + p_2 \\ \dddot{y} &= p_1g\dot{\alpha} + p_3 \\ y^{(4)} &= p_1g\ddot{\alpha} + p_4 = p_1g\frac{2g\sin(\alpha)\left(M + m\right) - ma\dot{\alpha}^2\sin(2\alpha)}{a\left(2M + m\left(1 - \cos(2\alpha)\right)\right)} + p_4 \\ &+ \frac{2p_1g\cos(\alpha)}{a\left(2M + m\left(1 - \cos(2\alpha)\right)\right)}F_{\rm C} =: a\left(\mathbf{x}, \mathbf{p}\right) + b\left(\mathbf{x}, \mathbf{p}\right) \cdot v \end{split}$$

	Example 3	Outlook	
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Approximate Transformation into Nonlinear Controller Canonical Form (2)

State-space representation for control design

- State vector after transformation of coordinates $\mathbf{x} = \begin{bmatrix} y & \dot{y} & \ddot{y} \end{bmatrix}^T$
- Control input $v := F_{\rm C}$
- Interval parameters $[p_i]$, $i \in \{1, \ldots, 4\}$ (representation of approximation and modeling errors)
- \bullet Error for angle measurement $[-0.01\ ;\ 0.01]\ rad$ and carriage position measurement $[-0.01\ ;\ 0.01]\ m$

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= x_3 \\ \dot{x}_3 &= x_4 \\ \dot{x}_4 &= a\left(\mathbf{x}, \mathbf{p}\right) + b\left(\mathbf{x}, \mathbf{p}\right) \cdot v \end{aligned}$$

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Violation of two-sided state constraint \implies Barrier function is **deactivated**, parameters and measured states are **uncertain**

	Example 3 0000●	Outlook 000	



No violation of two-sided state constraint \implies Barrier function is activated, parameters and measured states are uncertain

		Outlook ●00	

Outlook: Interval-Based Backstepping Control (1)

Interval-based backstepping control

• System model is now given in strict feedback form

$$\dot{x}_{1} = a_{1} (x_{1}, \mathbf{p}) + b_{1} (x_{1}, \mathbf{p}) \cdot x_{2}$$
$$\dot{x}_{2} = a_{2} (x_{1}, x_{2}, \mathbf{p}) + b_{2} (x_{1}, x_{2}, \mathbf{p}) \cdot x_{3}$$
$$\vdots$$
$$\dot{x}_{n} = a_{n} (x_{1}, \dots, x_{n}, \mathbf{p}) + b_{n} (x_{1}, \dots, x_{n}, \mathbf{p}) \cdot v$$

• Uncertain parameters $\mathbf{p} \in [\mathbf{p}]$

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Outlook: Interval-Based Backstepping Control (2)

Interval-based backstepping control

$$\begin{aligned} \dot{x}_1 &= a_1 \left(x_1, \mathbf{p} \right) + b_1 \left(x_1, \mathbf{p} \right) \cdot x_2 \\ \dot{x}_2 &= a_2 \left(x_1, x_2, \mathbf{p} \right) + b_2 \left(x_1, x_2, \mathbf{p} \right) \cdot x_3 \\ \vdots \\ \dot{x}_n &= a_n \left(x_1, \dots, x_n, \mathbf{p} \right) + b_n \left(x_1, \dots, x_n, \mathbf{p} \right) \cdot \end{aligned}$$

Control procedure

- Successive stabilization of the dynamics for x_i , $i = 1, \ldots, n-1$
- Treatment of x_{i+1} as a virtual control signal in the equation for \dot{x}_i
- Stabilization using a Lyapunov function candidate, e.g. $V_i=\frac{1}{2}x_i^2$, $\dot{V}_i=x_i\dot{x}_i<0$
- Differentiability of the virtual control signals is required up to x_n

		Outlook	
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Outlook: Interval-Based Backstepping Control (3)

Interval-based backstepping control

$$\dot{x}_{1} = a_{1} (x_{1}, \mathbf{p}) + b_{1} (x_{1}, \mathbf{p}) \cdot x_{2}$$
$$\dot{x}_{2} = a_{2} (x_{1}, x_{2}, \mathbf{p}) + b_{2} (x_{1}, x_{2}, \mathbf{p}) \cdot x_{3}$$
$$\vdots$$

$$\dot{x}_n = a_n \left(x_1, \dots, x_n, \mathbf{p} \right) + b_n \left(x_1, \dots, x_n, \mathbf{p} \right) \cdot v$$

Control procedure

- Successive stabilization of the dynamics for x_i , $i = 1, \ldots, n-1$
- Treatment of x_{i+1} as a virtual control signal for the equation \dot{x}_i
- Variable-structure approach for control signal v is possible (including sign(·)), for earlier stages smooth or regularized controllers (e.g. tanh(·))
- Finally: Proof of the overall system stability

		Outlook	Conclusions
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Conclusions and Outlook on Future Work

- Control-oriented modeling of dynamic systems
- Verified parameter identification as the basis for control design
- Stabilization of the error dynamics using interval techniques
- Handling of input and state constraints (guaranteed overshoot prevention, two-sided worst-case bounds for the system output)
- Use of interval analysis in real time

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Conclusions and Outlook on Future Work

- Control-oriented modeling of dynamic systems
- Verified parameter identification as the basis for control design
- Stabilization of the error dynamics using interval techniques
- Handling of input and state constraints (guaranteed overshoot prevention, two-sided worst-case bounds for the system output)
- Use of interval analysis in real time
- Extension by a sensitivity-based predictive controller
- Extension by a (sensitivity-based) state and disturbance observer
- Extension by backstepping-like control procedures

		Outlook	
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Merci beaucoup pour Votre attention! Thank you for your attention! Спасибо за Ваше внимание! Dziękuję bardzo za uwage! ¡Muchas gracias por su atención! Grazie mille per la vostra attenzione! Vielen Dank für Ihre Aufmerksamkeit!

A. Rauh et al.: Interval Methods for Robust Variable-Structure Control with One- and Two-Sided State Constraints